

## MINOR CHANGE APPROVAL

### 10078225

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### **S.E.A.M. SERVICE ELECTRONIC AVIATION MARINE S.A.R.L.**

**AERODROME DE MONTPELLIER L'OR  
34130 CANDILLARGUES  
FRANCE**

EASA.AP209

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** EASA.A.367

**Type Certificate Holder:** CEAPR

**Type:** DR 200, DR 300, DR 400 SERIES

**Model:** DR 400 NGL, DR 400 RP

DR 400/100, DR 400/120

DR 400/120 A, DR 400/120 D

DR 400/125, DR 400/125i

DR 400/140, DR 400/140 B

DR 400/160, DR 400/160 D

DR 400/180 S, DR 400/2+2

DR 400/180, DR 400/180 R

DR 400/200 I, DR 400/200 R

DR 400/500

**Description of Design Change: MFD AV-30C Installation (i.a.w EASA STC 10075856)**

See Continuation Sheet(s)

**For the European Union Aviation Safety Agency**

**Cologne, Germany, 28 January 2022**



**Marco CAPACCIO**  
Section Manager

**Small Aircraft, Balloons & Airships**



MFD AV-30C Installation (i.a.w EASA STC 10075856)

**EASA Certification Basis:**

The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

**Associated Technical Documentation:**

Technical Dossier AV 30C-DR 400 Revision 0 dated 26 January 2022

Or later revisions of the above listed document(s) approved/accepted under the EASA system.

**Limitations/Conditions:**

The approval holder shall fulfill the obligations of Part 21, Point 21.A.109.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -